

University of Illinois at Urbana-Champaign



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Air Conditioning and Refrigeration Center    A National Science Foundation/University Cooperative Research Center

## **Condensing Units for Household Refrigerator-Freezers**

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## I. Introduction

Refrigerator-freezer is a major household appliance designed for preserving foods through refrigerating and freezing. The increasing market demand for energy saving of the household appliance speed up to develop energy efficient system. The energy use of U.S. refrigerators has declined more than 60% in the last 25 years but changes in U.S. refrigerator minimum efficiency standards are reducing refrigerator energy use by around 25% below current levels [1]. To improve thermal efficiency of the system, the performance of the components constituting the refrigeration system should all be improved simultaneously. Reduction of thermal resistance of both condenser and evaporator is necessary in addition to higher compressor efficiency, improved characteristics of the expansion device, and better insulation of the enclosure. As a first step toward the eventual goal, the research project is focused on the condensing unit, which is composed of a compressor, a sub-condenser and a cooling fan.

The optimum arrangement of the condensing unit as well as the configuration of the sub-condenser is important to improve energy efficiency of the system, and reduce installation space and eventually increase available volume of refrigerator or freezer compartment. Improving thermal performance of the sub-condenser is required to reduce thermal resistance between refrigerant and environment. According to the previous research on wire-on-tube condensers, the external resistance for a refrigerator condenser is at least 95% of the total thermal resistance for the two-phase region and greater than 62% for the superheated and sub-cooled regions [2]. Therefore, decreasing the air-side thermal resistance is needed. To reduce this resistance, the surface area of the sub-condenser can be increased or the air-side heat transfer coefficient can be improved. However, the overall size of the sub-condenser is somewhat limited by economic and spatial constraints. Therefore, this project concentrated primarily on optimizing the air-side configuration of the sub-condenser.

In addition, reasonable performance data of the compressor with given operation condition is needed for system designer to make optimal design of the system. Compressor manufacturers usually provide empirical performance curves called compressor maps, expressing mass flow and power input as polynomial functions of evaporation temperature for a range of condensing temperatures. Since they are based on fixed ambient and suction gas temperatures, the maps are useful for comparing and selecting the compressors. However, they are inadequate for general system analysis, because they contain no information about the effects of different ambient and suction temperatures and are unable to predict discharge temperatures, which define the condenser inlet condition. There are several detailed compressor models in the open literature, but they usually require large sets of unknown parameters to characterize the complicated refrigerant flow, heat exchanges, effects of oil, moving boundaries, and data on the geometry and heat transfer characteristics of internal parts. Most such models contain too little data to distinguish accurate parameter estimation from merely "tuning" the model for a few selected operating conditions. The research project is focusing on this issue in order to establish a foundation for the development of a simple physical model of small hermetic reciprocating compressors, suitable for use in the design of refrigerator-freezers. Unlike the polynomial functions in compressor maps, it may be safer to extrapolate physically-based models outside the range of compressor calorimeter operating capabilities.

## II. Compressor Analysis

The basic purpose of the compressor sub-model in a larger system model is to provide the refrigerant mass flow rate, power consumption and discharge states of the compressor using some given information. The usual input data are compressor geometry (displacement and clearance volume), compressor speed, suction pressure and temperature, discharge pressure, and ambient temperature.

A mass flow rate model reflects clearance volumetric efficiency and simulates suction gas heating using an empirical relationship: the specific volume at the suction port of the cylinder has linear relationship with the specific volume at the compressor suction. Compressor work is calculated using the compressor efficiency represented by some empirical parameters. A linear relationship between the discharge and shell temperatures will be used for calculating the discharge temperature.

The goal is to develop a general but simple model for reciprocating compressors for refrigerator-freezers using R-134a refrigerant, to provide reasonable compressor performance data to system designer. The basic simple physical model [3] was developed based on thermodynamic principles and large data sets from compressor calorimeter and *in-situ* tests [4, 5].

### 1. Introduction

The compressor is one of major components of an air conditioning or refrigeration system and has an important effect on system energy efficiency. Compressor manufacturers usually provide empirical performance curves called compressor maps, expressing mass flow and power input as polynomial functions of evaporation temperature for a range of condensing temperatures. Since they are based on fixed ambient and suction gas temperatures, the maps are useful for comparing and selecting the compressors. However, they are inadequate for general system analysis, because they contain no information about the effects of different ambient and suction temperatures and are unable to predict discharge temperatures, which define the condenser inlet condition. Dabiri and Rice [6] suggested additional assumptions to account for suction gas heating, which have been used with limited success for reciprocating compressors with low-side sumps, and with greater success for rotary compressors where the suction gas is injected directly [7]. Haider et al. [8] investigated the effect of compressor map and ambient temperature on the power consumption of a refrigerator-freezer using the ERA (EPA Refrigerator Analysis) software, complemented by the measured compressor and refrigerator-freezer data. They reported the accuracy of ERA estimation could be improved up to 5.1% at an ambient temperature of 43.3°C by using the measured map data at 43.3°C rather than the given map at standard test condition of 32.2°C.

There are several detailed physical models of compressors in the literature. Prakash and Singh [9] developed a mathematical model for a reciprocating compressor using the first law of thermodynamics and assuming the refrigerant is an ideal gas. Hiller and Glicksmann [10] reported a detailed compressor model of reciprocating compressors and compared the simulation results with experimental data. Domanski and Didion [11] developed a quite detailed compressor model for a system simulation, but it requires over 30 input parameters. Todescat et al. [12] presented a thermal energy analysis of a reciprocating hermetic compressor using the energy balances for several parts of the compressor. They reported the effect of compressor shell temperature on the compressor performance such as power consumption and energy efficiency ratio. Recently, Cavallini et al. [13]

presented a steady state model for the thermal analysis of an hermetic reciprocating compressor and compared with a few experimental data points for R-134a and R-600a compressors. Rigola et al. [14] presented an advanced numerical scheme of the thermal and fluid dynamic behavior of small hermetic reciprocating compressors, along with some empirical data. Those models usually require large sets of unknown parameters to characterize the complicated refrigerant flow, heat exchanges, effects of oil, and moving boundaries, plus data on the geometry and heat transfer characteristics of internal parts. Most such papers contain too little data to distinguish accurate parameter estimation from merely "tuning" the model for a few selected operating conditions. Klein and Reindl [15] developed a model similar to that presented here, which embodies different physical assumptions underlying the mass flow and power submodels.

The purpose of this study is to extend to different type of reciprocating compressors on approach explored first for small hermetic reciprocating compressors [3] and provide the reasonable compressor data to system designer. Instead of taking a detailed approach, the goal is to find the simplest formulation suitable for use in refrigerator model. Pressure losses along the refrigerant path are neglected and the compression process is assumed as isentropic. A mass flow rate model reflects clearance volumetric efficiency and simulates suction gas heating using an empirical relationship: the specific volume at the suction port of the cylinder has linear relationship with the specific volume at the compressor suction. Compressor work is calculated using the compressor efficiency represented by only two empirical parameters. A linear relationship between the discharge and shell temperatures is extracted from the data and used for calculating the discharge temperature for any ambient air temperature.

## **2. Simple compressor model**

The basic purpose of the compressor sub-model in a larger system model is to provide the refrigerant mass flow rate, power consumption and discharge states of the compressor using some given information. The usual input data are compressor geometry (displacement and clearance volume), compressor speed, suction pressure and temperature, discharge pressure, and ambient temperature.

A detailed thermodynamic model of a compressor is extremely complex due to the inherently complicated structure of the compressor and refrigerant flow pathways, along which heat transfer and pressure vary substantially and rapidly. For some applications (e.g. design of variable-speed drives) such detailed modeling is necessary [16]. Our hypothesis is that the requirements for quasi-steady system simulation modeling are much less demanding, so several assumptions can be made to simplify the physical model for small hermetic reciprocating compressors:

1. The refrigerant path can be treated as a steady state flow;
2. The compression process is isentropic;
3. The kinetic and potential energies of refrigerant are neglected;
4. The pressure losses along the refrigerant path are neglected;
5. The oil effects on the refrigerant properties are neglected.



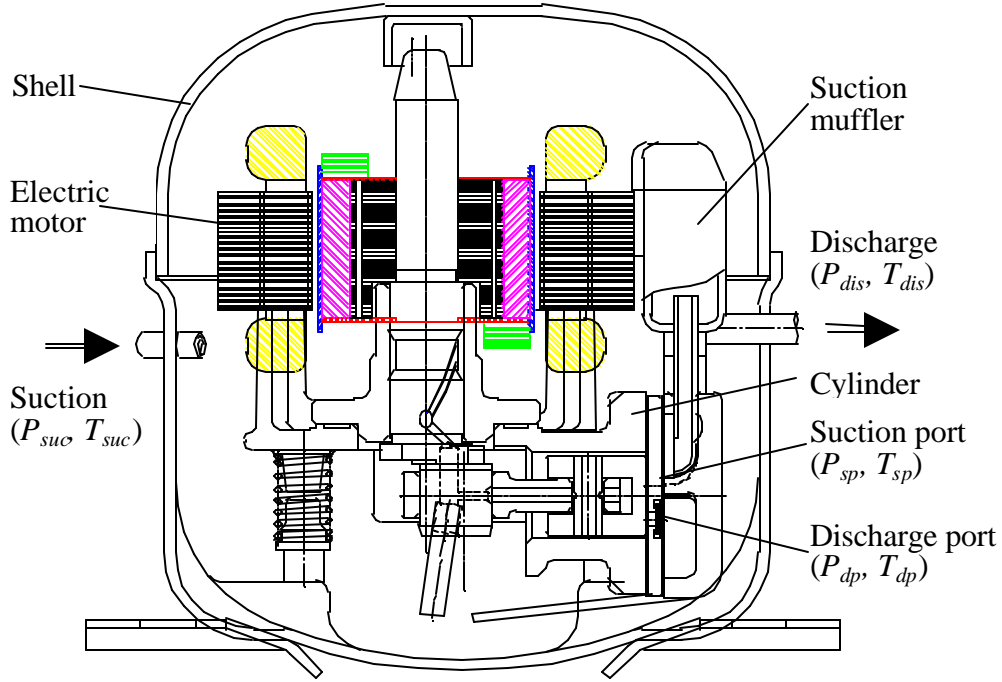


Fig. 1 Schematic diagram of a reciprocating compressor

Fig. 1 shows a schematic diagram for typical low-side sump reciprocating compressor. The following is a simple description for the refrigerant compression process of a low-side sump reciprocating compressor. Refrigerant vapor at low pressure and temperature enters the compressor shell through suction line and is heated as it cools the motor and other parts, and mixes with hot plenum gas along the refrigerant suction path. After compression in the cylinder, high temperature and pressure refrigerant gas discharges through a muffler, rejecting heat to the plenum gas before exiting via the discharge line. Therefore, for the low-side sump compressors most of the shell is at low suction pressure.

For a given compressor velocity and swept volume, the mass flow rate can be calculated using the volumetric efficiency [17].

$$\dot{m} = 60 \frac{h_v V_{disp} \omega}{v_{sp}} \quad (2.1)$$

The volumetric efficiency for reciprocating compressors is given by Eq. (2.2), which accounts for re-expansion of the gas remaining in the clearance volume.

$$h_v = 1 - C \left( \frac{v_{sp}}{v(P_{dp}, s_{sp})} - 1 \right) \quad (2.2)$$

For a small hermetic compressor, we assume compression and re-expansion processes to be isentropic so the specific volume at the discharge port of the cylinder can be calculated. The decrease in density due to suction gas heating and mixing between the suction line to suction port has a large effect on the volumetric efficiency and mass flow rate.

For small hermetic reciprocating compressors considered here, the following empirical relations are obtained using least squares from the mass flows and suction pressure (Fig. 3)

$$v_{suc} = c_1 + \frac{c_2}{P_{suc}} \quad (2.3)$$

$$v_{sp} = c_3 + \frac{c_4}{P_{suc}}$$

where  $c_1$ ,  $c_2$ ,  $c_3$ , and  $c_4$  are constants to be determined.

Rearranging the Eq. (2.3) shows relationship between the specific volume for the suction and suction port

$$v_{sp} = a_1 + a_2 v_{suc} \quad (2.4)$$

where the constants can be estimated directly from the same experimental data.

The constants of Eqs. (2.3) and (2.4) are determined by minimizing the following objective function to obtain the best agreement with the measured mass flow rate at  $N$  data points.

$$obj\_m = \min \sqrt{\frac{\sum_{n=1}^N \left( \frac{\dot{m}_{exp} - \dot{m}_{cal}}{\dot{m}_{exp}} \right)^2}{N}} \quad (2.5)$$

The compressor power consumption can be calculated if the compressor efficiency is known. The isentropic compressor efficiency is normally defined across the entire compressor shell, but in our case we define a compression efficiency that excludes the effects of subsystem heat transfers upstream of the suction port, and includes the effects of motor efficiency:

$$h_c = \dot{m} [h(P_{dp}, s_{sp}) - h_{sp}] / W \quad (2.6)$$

Empirical observation (Fig. 5) shows that the compressor efficiency can be expressed simply as:

$$h_c = k_1 + k_2 T_{dis} \quad (2.7)$$

where the constants  $k_1$  and  $k_2$  can be estimated by minimizing the average normalized deviation between measured and calculated power consumption

$$obj\_W = \min \sqrt{\frac{\sum_{n=1}^N \left( \frac{W_{exp} - W_{cal}}{W_{exp}} \right)^2}{N}} \quad (2.8)$$

The empirically observed dependence of compression efficiency on shell temperature is thought to reflect the temperature dependence of oil viscosity. The principal mechanism of oil cooling, splattering into the compressor shell, determines the relationship between shell temperature and oil temperature.

Three more parameters are needed to estimate the discharge temperature. The first step is to apply the first law of thermodynamics across the compressor shell using the following equation for the steady state flow, neglecting the potential and kinetic energy.

$$Q = W - \dot{m}(h_{dis} - h_{suc}) \quad (2.9)$$

The previous experimental observations [3] suggested a linear relationship with the discharge temperature as shown in Fig. 7

$$T_{shell} = a + bT_{dis} \quad (2.10)$$

where  $a$  and  $b$  are empirical constants determined from at least two experimental data points.

Finally, the heat transfer from the compressor shell can be obtained from the equation

$$Q = UA_{shell}(T_{shell} - T_{amb}) \quad (2.11)$$

If discharge temperatures are available for more operating conditions,  $UA_{shell}$  is a constant to be determined using the least squares method with the measured heat loss values

$$obj\_Q = \min \sqrt{\frac{\sum_{n=1}^N \left( \frac{Q_{exp} - Q_{cal}}{Q_{exp}} \right)^2}{N}} \quad (2.12)$$

A Newton-Raphson based equation solver [18] was used for both the simulation and optimization calculations needed to estimate these parameters.

### 3. Simulation and parameter estimation results

Table 1 and Appendix A show data sets used in this study. Data sets for three R-134a compressors (DK172B, DK190B, ZK180) are used: data sets I, II and III consist of 20, 16 and 16 data points, respectively, from calorimeter tests conducted at 32.2°C ambient over a wide range of suction and discharge pressures. In case of data set II, some data points for the shell temperature were missing, so only nine data points were used for determining the constants  $a$  and  $b$  from Eq. (2.10). The cylinder volumes for each compressor are 7.2, 9.0, and 8.0 cc, respectively and the nominal compressor speed is 3450 rpm at 60 Hz. The clearance volume was unknown, so those values were estimated simultaneously using the mass flow rate submodel at the compressor speed of 3450 rpm as shown in Table 1.

Table 1. Data sets used for computer simulations

Data set	Compressor Model	Displacement/ Clearance (cc)	Refrigerant	Test methods	Number of data points (N)
I	DK172B	7.2/0.175	R-134a	compressor calorimeter	20
II	DK190B	9.0/0.214	R-134a	compressor calorimeter	16
III	ZK180	8.0/0.200	R-134a	compressor calorimeter	16

The simulation results are depicted in Figs. 3–8, compared with measured values. Tables 2 and 3 show the estimated parameters and RMS errors of simulation results for each data set, respectively. The parameter estimates were obtained using the complete data sets, except missing data; the small standard deviations apparent from Figs. 3-8 indicate that smaller subsets of data could yield similar results. The overall calculation results are in good agreement with measured data within a reasonable accuracy (especially considering measurement accuracy) as shown in Table 3: the RMS errors for calculated mass flow rates and power inputs are within 2.5% and 2.9%, and the difference between the calculated and measured discharge temperatures is below 3.8°C.

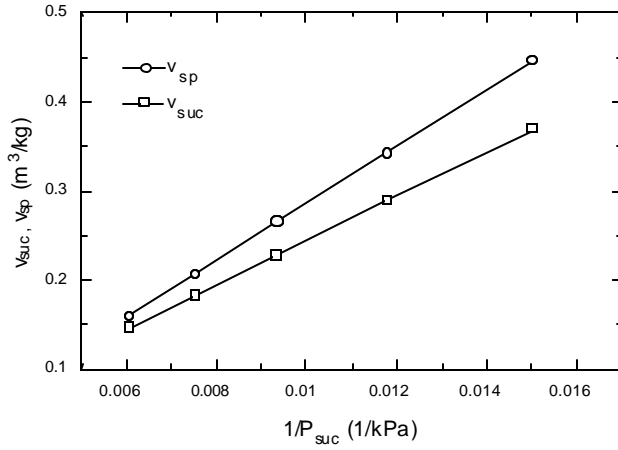
Table 2. Estimated parameters

Data set	Estimated constants						
	$c_3$	$c_4$	$k_1$	$k_2$	$a$	$b$	$UA_{shell}$
I	-0.03135	31.85	0.367	0.00289	43.77	0.255	3.69
II	-0.00483	27.23	0.255	0.00343	26.27	0.474	3.65
III	-0.00652	26.65	0.315	0.00336	46.48	0.248	3.57

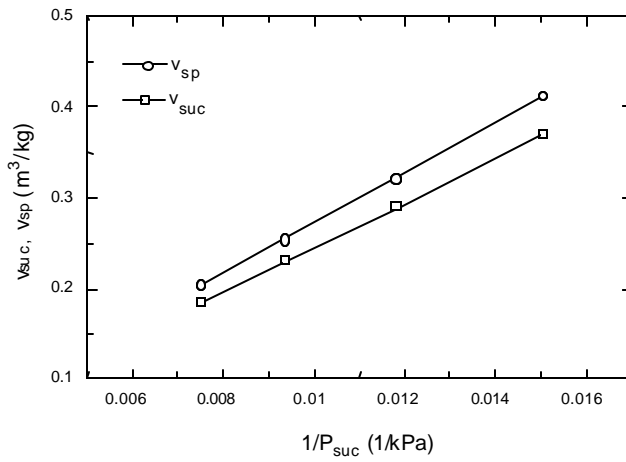
Table 3. RMS errors

Data set	RMS errors		
	%		°C
	$obj_{mr}$	$obj_W$	$DT_{dis}$
I	2.50	0.52	3.5
II	1.47	2.85	3.8
III	1.48	2.10	3.7

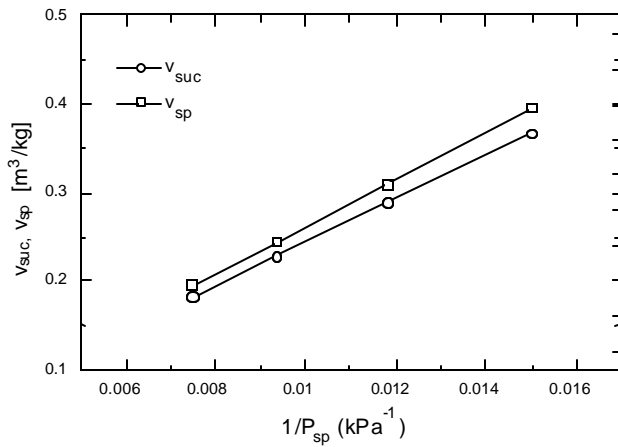
Figs. 2-4 present the simulation results of the mass flow rate submodel. Fig. 2 shows the linear relationships between the specific volumes at the suction and suction ports and the inverse of suction pressure. The specific volumes at the suction port are consistently larger than those at the suction, suggesting that the suction gas is heated as it cools the motor and other parts and mixes with hot plenum gas along the refrigerant suction path. The suction gas heating and mixing cause density decrease, which is a very important factor affecting mass flow rate. Fig. 3 presents the clearance volumetric efficiency. As expected, the volumetric efficiency decreases systematically with the pressure ratio. It varies from 0.9 to 0.6 with increasing pressure ratio of 5 to 20, and has a value of about 0.83 at the normal operating condition where the pressure ratio is around 10. Fig. 4 presents a comparison of the calculated and measured mass flow rates and they are in excellent agreement over a whole range of test conditions.



(a) DK172B



(b) DK190B



(c) ZK180

Fig. 2 Specific volumes as function of suction pressure

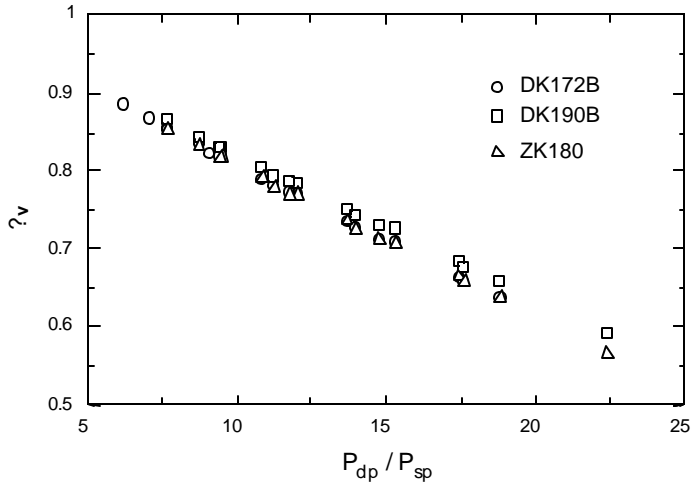


Fig. 3 Volumetric efficiency

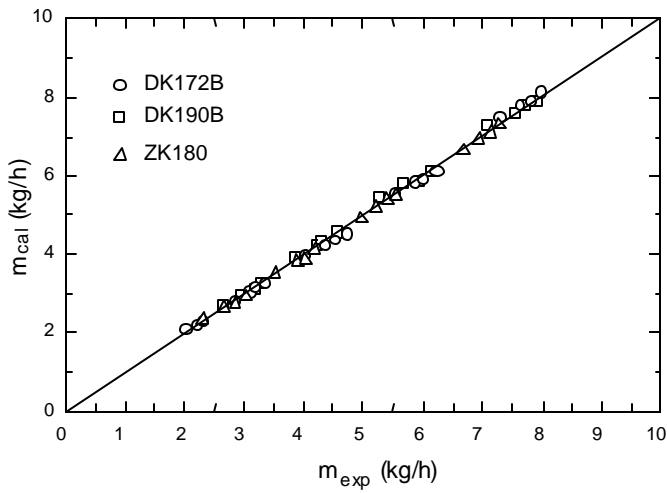


Fig. 4 Refrigerant mass flow rate

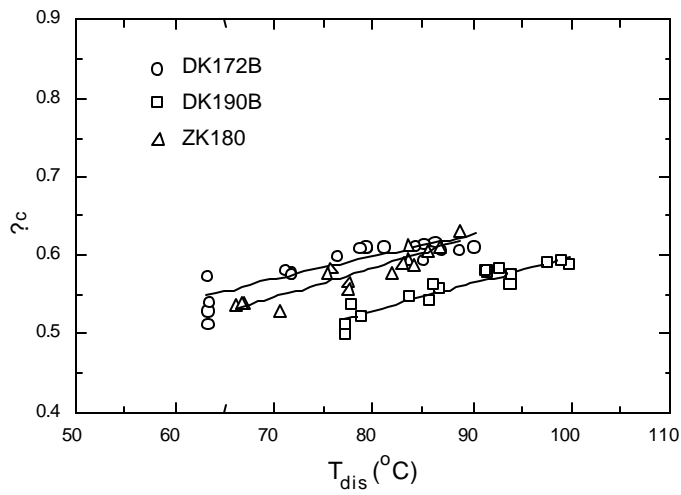


Fig. 5 Compressor efficiency as a function of discharge temperature

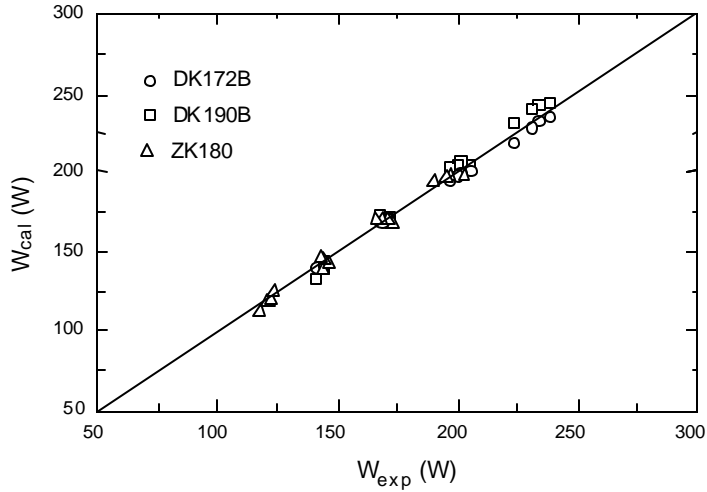


Fig. 6 Power consumption

Figs. 5 and 6 present the results of power consumption submodel. Fig. 5 depicts compressor efficiency. Figs. 7 and 8 show comparison of the calculated and measured discharge temperatures. Fig. 7 demonstrates the linearity of the empirical correlation between the compressor shell and discharge temperatures. This relation is used for estimating the discharge temperature and the overall heat transfer coefficient,  $UA_{shell}$ , of the compressor shell using Eq. (2-11). The estimated values of all the parameters affect the calculated discharge temperatures, which are compared to measured data in Appendix A and Fig. 8.

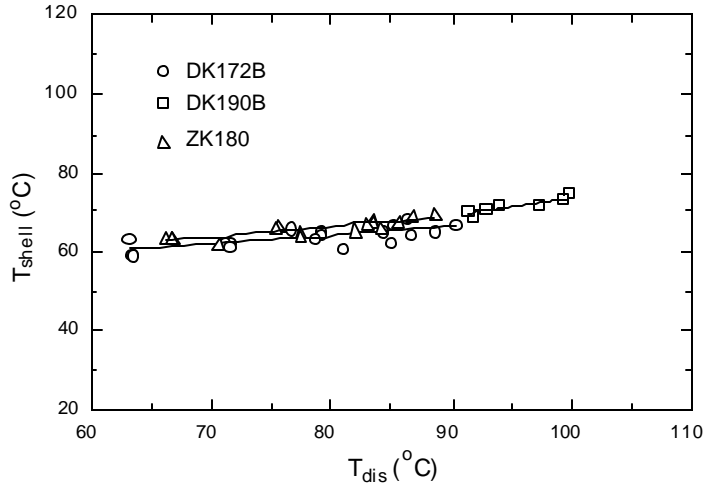


Fig. 7 Compressor shell temperature as a function of discharge temperature

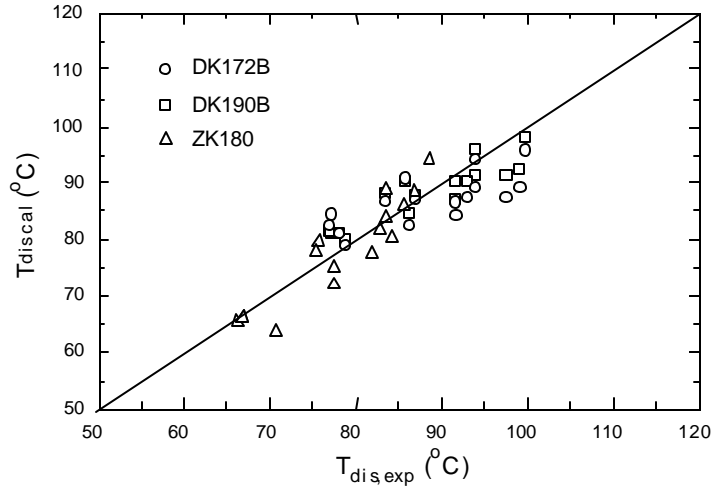


Fig. 8 Discharge temperature

#### 4. Conclusions

A simple and general compressor model developed using the thermodynamic principles and large data sets from the compressor calorimeter and *in-situ* tests has been successfully applied to the different type of reciprocating compressors. The model can estimate mass flow rate and compressor power consumption with rms errors less than 3.0%, which is not much larger than measurement errors associated with calorimeter testing under ideal conditions. The magnitude of the errors suggests that the seven parameters could be estimated from data sets much smaller than those used here. The density decrease due to suction gas heating and mixing is a very important factor affecting mass flow rate. Oil viscosity appears to be the most important factor accounting for compressor power consumption more than the isentropic ideal.

Fewer tests should be needed in calorimeter to estimate these seven physical parameters, compared to the 18-20 parameters needed for today's polynomial curve fits of mass flow and power. However this approach requires that  $T_{shell}$  and  $T_{dis}$  should be recorded for at least two operating conditions during the calorimeter tests, in order to define the simple linear relation between them and to estimate the compressor shell heat transfer coefficient. Similarly, only two such data points would need to be obtained by OEM's to characterize accurately the compressor shell heat transfer coefficient in unique installations, where the air temperature at the compressor is influenced by the location of the condenser.



### III. Sub-Condenser Design

The sub-condensers considered are heat exchangers; hence, the continuously changing temperature difference between air and refrigerant streams should be accounted for the performance analysis of the condenser. For the several different configurations of the condensers, the effectiveness- $NTU$  method is used for the analysis.

#### 1. Condenser Simulation Models

##### 1.1 Introduction

The following sections describe the development of simulation models for three advanced condenser geometries:

1. Vertical cross-counterflow,
2. Spiral cross-counterflow,
3. Sawtooth cross-parallel flow.

The first parts describe the methods used to simulate the thermal-hydraulic performance of these condensers. Next the simulation results are compared with the manufacturer's experimental results [19] to validate the simulation models. Finally an optimization is conducted for the sawtooth cross-parallel flow condenser, by combining the condenser and compressor submodels, and searching over four degrees of freedom to identify the COP-maximizing combination of wire and tube diameters and spacings, subject to the design constraints provided by manufacturer [19].

##### 1.2 Condenser geometry

Wind tunnel experiments were conducted on different condensers: cross-counterflow (C/F), spiral and sawtooth (S/T). The specifications of these condensers are listed in Table 1.1. When performing experiments, these condensers were placed in ducts. A sketch of the duct is shown in Fig. 1.1. This duct has three rows of holes that allow air to flow from bottom. A portion of air enters the duct from the front of the duct and as it flows over the coil, incoming air from the bottom rows of holes gets mixed with it, the pressure differential of air-flow providing the suction force needed to pull the air from the bottom holes. Based on their measurements, manufacturer estimated that about 30% of the total air enters the duct from the front and the remaining 70% of the total enter through the bottom of duct [19]. In the wind tunnel experiments, the air entrance holes on either side of the duct were closed. The downwind end of the duct was mostly closed, except for the 121 mm gap where the fan would be positioned, which guided the outgoing air to the nozzle for air-flow measurements. To simulate this unique air-flow arrangement, basic finite element simulation model was developed, as described in the next few sections.

Table 1.1 Specifications of condensers

	ET-Basic	C/F#1	C/F#2	C/F#3	Spiral#1	Spiral #2	Spiral#3	S/T#1	S/T#2
Leg length (mm)	-	465	465	465	410	410	410	468	468
Tube pitch (mm)	20	20	20	20	30	35	40	20	20
Row	9	2	2	2	1	1	1	1	1
Step	10	19	18	18	12	10	8	24	24
Fin number	260	1596	1512	2088	908	764	613	86	122
Fin thickness (mm)	1.6	1.6	1.6	1.6	0.35	0.35	0.35	1.6	1.6
Fin pitch (mm)	10(5)	10	10	7	6	6	6	10	7
Fin length (mm)	-	33	33	33	-	-	-	474	474
Tube length (m)	15.51	19.47	18.49	18.49	5.47	4.73	3.88	11.93	11.93
Tube O.D.(mm)	4.76	4.76	4.76	4.76	4.76	4.76	4.76	4.76	4.76
Tube thickness (mm)	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7	0.7
Heat transfer area(m <sup>2</sup> ) (air/ref-side)	0.464/ -	0.562/ 0.205	0.533/ 0.195	0.631/ 0.195	1.037/ 0.058	0.898/ 0.050	0.735/ 0.041	0.384/ 0.126	0.470/ 0.126
Duct height (mm)	-	40	40	40	25	25	25	35	35

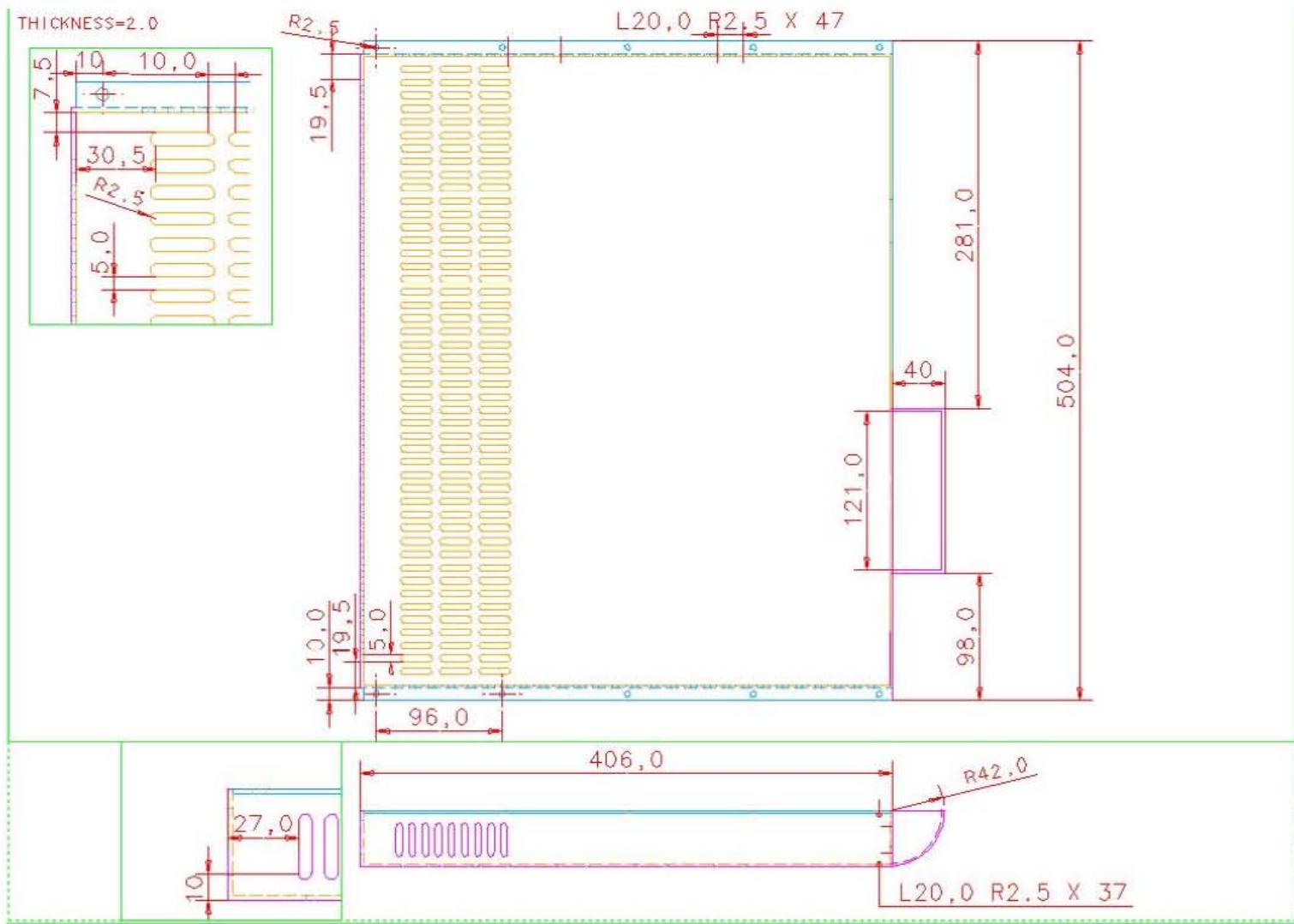


Fig. 1.1 Duct geometry

### 1.3 Simulation models

Different simulation models were developed for the three different condensers. The vertical cross-counterflow and spiral type condensers have similar features. The cross-parallel flow sawtooth coil simulation has variations and is described separately.

#### 1.3.1 Cross-counterflow condensers

A vertical cross-counterflow coil and air-flow arrangement is shown in Fig. 1.2 to illustrate the features: a number of identical elements consisting (in this case) of two tube passes and 84 wires welded to the front and back of the tubes.

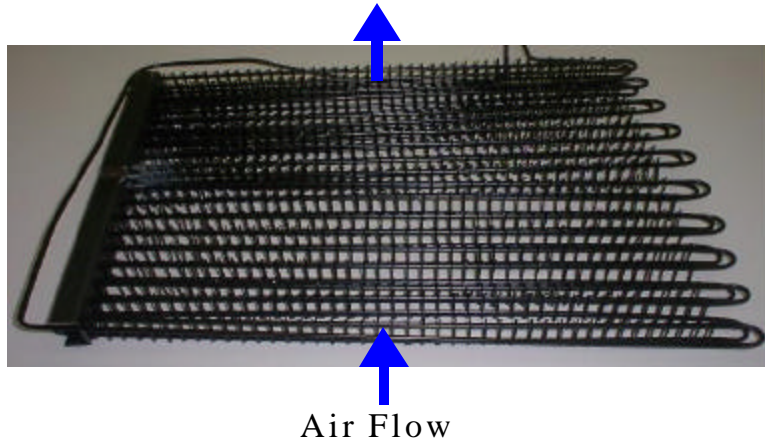


Fig. 1.2 Vertical Cross-counterflow coil and Air Flow

This was divided into a number of finite element divisions equal to the number of layers (slabs) in the coil. Each element was modeled as a cross-flow heat exchanger using the effectiveness-NTU method:

$$e = \frac{Q_{\text{actual}}}{Q_{\text{max}}} = \frac{Q_{\text{actual}}}{C_{\text{min}} \Delta T_{\text{max}}} = \frac{Q_{\text{actual}}}{C_{\text{min}} (T_{\text{r,in}} - T_{\text{a,in}})} \quad (1.1)$$

$C_{\text{min}}$  is the minimum heat capacity between refrigerant and air (for some single phase elements,  $C_{\text{min}}$  corresponds to the refrigerant stream heat capacity). For specific heat exchanger conditions (i.e. geometry, type of flow) the effectiveness can be related to the size of the heat exchanger through the number of transfer units (NTU).

$$\text{NTU} = \frac{UA}{C_{\text{min}}} \quad (1.2)$$

By using the effectiveness-NTU method and providing an initial guess for the refrigerant outlet state, it is possible to simulate the entire condenser by solving the hundreds of equations in a sequential manner. This series of equations is contained in a single procedure, which employs the upstream marching algorithm suggested by Harshbarger and Bullard [20]. The refrigerant exit state is passed to this procedure from the main program. From this condenser exit state, the inlet state of the refrigerant to that element is obtained. The effectiveness-NTU method

is used to determine the amount of heat transfer in that element. The refrigerant pressure drop is determined within the element. Thus, the refrigerant properties at the exit from the upstream element are obtained.

Marching upstream in this fashion, the refrigerant properties ( $T_{dis}$ ,  $P_{dis}$ ) at the inlet to the condenser are calculated and returned to the main program. If the calculated refrigerant state is not exactly equal to the actual refrigerant inlet state, the above steps are repeated using some updated condenser exit state, after executing a Newton-Raphson algorithm in the main program using Engineering Equations Solver [18]. If there is phase change in an element from superheated to two-phase zone, the element is broken into two elements; the lengths of superheated and two-phase zones within that element are determined using single 1-D Newton-Raphson algorithm simulated within the procedure. The heat transfer and refrigerant pressure drop, etc. are then determined accordingly. For detailed description of this algorithm, refer to [20].

The number of elements where air enters from the bottom is calculated using:

$$Num_{elm,hole} = (Num_{elm} * Depth_{hole} / Depth) \quad (1.3)$$

where

$Num_{elm,hole}$  = number of element allowing air-flow from bottom,

$Num_{elm}$  = total number of elements (= number of slabs),

$Depth_{hole}$  = depth of duct from front to the point where the last row of holes end,

$Depth$  = total duct depth.

If this number is non-integer, it is rounded to the next higher integer. The amount of air entering from the front of the duct is assumed to be 30% of total (as specified in [19]). The remaining 70% air from the bottom is equally divided among the number of elements having holes at the bottom.

The air-side pressure drop across each element is multiplied by the number of elements and the total air-side pressure drop is computed. The duct pressure drop of the rectangular duct is calculated and added. The calculations neglect the pressure drops across the front grille and, due to turning of air to converge at the hole where fan is located (Fig. 1.1). The simulations also neglected the bypass effect of air flowing over the return bends where no wires are present.

The refrigerant pressure drop in each element is added up to obtain the total pressure drop. Usually, the pressure drop in two-phase forms the largest component of the total pressure drop. The entrance and exit lengths of the tube were neglected in the pressure drop computations and also in heat transfer, because, they formed a small fraction of the total tube length (1 m in 19 m for C/F#1 coil). Similarly, the total heat transfer is obtained by summing the heat transfer occurring in each element.

For wire-on-tube condensers, wires act as fins, so the fin efficiency ( $\eta_w$ ) must be taken into account. In Hoke, et al [21], it has been shown that,

$$\eta_w = \frac{\tanh m}{m} \quad (1.4)$$

$$m^2 = \frac{h_w S_t^2}{k_w D_w} \quad (1.5)$$

The correlations used to determine the value of  $h_w$  are geometry-specific and therefore are given separately for each condenser. The surface resistance ( $R_{surf}$ ) is given by

$$R_{surf} = \frac{1}{A_{eff} h_w} \quad (1.6)$$

where

$$A_{eff} = \frac{A_t}{\sqrt{D_t/D_w}} + ?_w A_w \quad (1.7)$$

The overall UA is given by

$$UA = \frac{1}{R_{surf} + R_r} = \frac{1}{\frac{1}{A_{eff} h_w} + \frac{1}{A_{t,i} h_r}} \quad (1.8)$$

The air-side correlations is calculated from Nusselt number as a function of Reynolds number. Where

$$Nu_w = \frac{h_w D_w}{k_w} \quad (1.9)$$

and  $k_w$  was set to 60.5 W/m-K for steel wires.

In two-phase, effectiveness is related to NTU by,

$$e = 1 - e^{-NTU} \quad (1.10)$$

Also from Incropera and DeWitt [22], for single phase, the effectiveness for heat exchanger in cross-flow with both fluids unmixed, is given by,

$$\varepsilon = 1 - \exp\left[\left(\frac{1}{C_{rat}}\right)(NTU)^{0.22}\left\{\exp\left[-C_{rat}(NTU)^{0.78}\right]-1\right\}\right] \quad (1.11)$$

The single-phase refrigerant-side heat transfer coefficient is calculated by Dittus-Boelter [23]. The correlation developed by Dobson and Chato [24] for low mass fluxes is used to calculate the two-phase refrigerant heat transfer coefficient. Refrigerant-side pressure drop in the two-phase region is calculated using Souza and Pimenta [25], and the Darcy friction factor [26] in the single-phase (subcooled and superheated) region.

Thus, the common features to all cross-counterflow condensers have been discussed. The next subsections discuss the specific features of simulations of vertical layer and spiral condensers. The simulation results are compared with the experimental results.

### 1.3.1.1 Vertical cross-counterflow condenser

For this condenser, the air-side heat transfer and pressure drop is computed by the correlation developed by Lum and Clausing [27] with the angle of attack  $90^\circ$ . The air-side heat transfer coefficient is computed using equation (1.12)

$$Nu_w = C * Re_{\max}^{0.5744}$$

$$C = 0.502 \sin(a) \exp(-1.014a + 0.3775a^2) \quad \frac{p}{4} \leq a \leq \frac{p}{2} \quad (1.12)$$

The air-side pressure drop across the condenser is computed using equation (1.13)

$$\Delta P_{a,c} = C_D \left( \frac{1}{2} \rho_a V_{\max}^2 \right)$$

$$C_D = D_1 + D_2 Re_{\max}^{-0.06533}$$

$$D_1 = -0.7856 \sin(\alpha) \exp(1.177\alpha - 0.3229\alpha^2) \quad \frac{p}{4} \leq a \leq \frac{p}{2}$$

$$D_2 = 2.451 \sin(\alpha) \exp(0.2858\alpha) \quad \frac{p}{4} \leq a \leq \frac{p}{2}$$

$$Re_{\max} = \frac{\rho V_{\max} D_w}{\mu}$$

$$\frac{A_{\text{face}}}{A_{\text{min}}} = \frac{V_{\max}}{V_{\text{face}}} \equiv V_{\text{ratio}}$$

$$V_{\text{ratio}} = \frac{1}{\left\{ 1 - \frac{D_w}{S_w} - \frac{D_t (S_w - D_w)}{S_w S_t} \right\}} \quad (1.13)$$

Wind tunnel experiments were conducted by Park et al. [19] for the test conditions listed in Table 1.2. The results obtained for this vertical cross-counterflow condenser are compared with the experimental results in Tables 1.3 thru 1.5. Air flow rates greater than  $1 \text{ m}^3/\text{min}$  result in face velocities exceeding  $1 \text{ m/s}$  (through the front grille), which could create problems related to dust entrainment. Also at high flow rates, fan noise becomes a potentially serious problem. Note, however, in the simulations, that the refrigerant exit temperature difference was pinched at the two higher air-flow rates. Therefore, the higher air-flow rate simulations have not been conducted/reported. The experimental results confirm that only  $170 \text{ W}$  (calculated using  $(T,P)_{\text{dis}} = (65^\circ\text{C}, 1030 \text{ kPa})$  and  $(T,P)_{\text{exit}} = (30^\circ\text{C}, 1030 \text{ kPa})$  and refrigerant mass flow rate =  $3 \text{ kg/h}$ , neglecting refrigerant-side pressure drop) heat can be extracted from the refrigerant by increasing air flow, due to pinching at the exit. Adding either area (A) or enhancing heat transfer

(h) is not going to be helpful, because pinched condition implies that  $hA \rightarrow \infty$  and the heat capacity of the refrigerant stream has been reached. The measured air-side pressure drop is lower than what was predicted by the simulations. This was not expected, since the simulations include only the duct and condenser pressure drop and neglect the pressure drops due to front-grille and due to turning to converge at the hole where fan is located (Fig. 1.1). The experimental uncertainty of the air-side pressure drop measurements is not known.

The general trend of the results is that the capacity results are overpredicted by around 5% by the simulation model. Air flow nonuniformities caused by the blockage at the rear of the duct, neglected in the simulations, may account for a significant but unknown fraction of this difference. Uncertainty calculations in experimental results can be found in Appendix B.

Table 1.2 Test conditions

Parameters	Conditions
Air inlet temperature (°C)	30
Refrigerant inlet temperature (°C)	65
Refrigerant flow rate (kg/h)	3.0
Refrigerant inlet pressure (kPa)	1030

Table 1.3 Vertical cross-counterflow simulation Results(C/F#1)

Parameters	Results							
	Computed				Experimental			
Air flow Rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref-side)	113	145	171		108/ 111	143/ 139	166/ 165	170/ 166
$\Delta P_{air}$ (mmH2O)	0.50	0.96	1.80		0.34	0.67	1.44	2.57
$\Delta P_{ref}$ (kPa)	12	10	7.4		*	*	*	*
Exit quality	0.33	0.11	-		0.35	0.14	-	-
$T_{ref,cond,out}$ (°C)	40	40.1	30		40	40	35	33.7
$T_{air,cond,out}$ (°C)	38.6	37.7	36.1		*	*	*	*

- sub-cooled condenser exit

\* data not available



Table 1.4 Vertical cross-counterflow simulation results(C/F#2)

Parameters	Results							
	Computed				Experimental			
Air flow rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref-side)	110	142	170		104/ 108	138/ 138	163/ 165	165/ 167
$\Delta P_{air}$ (mmH <sub>2</sub> O)	0.50	0.93	1.75		0.33	0.66	1.46	2.52
$\Delta P_{ref}$ (kPa)	11	9.6	7.9		*	*	*	*
Exit quality	0.35	0.13	-		0.37	0.16	-	-
T <sub>ref,cond,out</sub> (°C)	40.1	40.1	30.3		40	40	35	33.6
T <sub>air,cond,out</sub> (°C)	38.5	37.5	36.1		*	*	*	*

- sub-cooled condenser exit

\* data not available

Table 1.5 Vertical cross-counterflow simulation results(C/F#3)

Parameters	Results							
	Computed				Experimental			
Air flow rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref-side)	124	159			112/ 117	143/ 144	164/ 165	166/ 166
$\Delta P_{air}$ (mmH <sub>2</sub> O)	0.58	1.07			0.44	0.81	1.77	3.1
$\Delta P_{ref}$ (kPa)	10.3	9.3			*	*	*	*
Exit quality	0.25	-			0.31	0.11	-	-
T <sub>ref,cond,out</sub> (°C)	40.1	39			40	39.4	35	34
T <sub>air,cond,out</sub> (°C)	39.4	38.4			*	*	*	*

- sub-cooled condenser exit

\* data not available

### 1.3.1.2 Spiral condenser simulation

The spiral condenser is very similar to the vertical cross-counterflow condenser as described in section 1.1, only difference being that the wires are replaced by spirally coiled fins. Hence, the simulation model is structurally similar. The air-side heat transfer coefficient developed by Kim [28], as listed in equation (1.14) below, was used in the simulation model.

$$Nu = \frac{h_o D_h}{k} = 0.414 Re^{0.551} \left( \frac{F_p - F_{th}}{F_h} \right)^{0.763} \left( \frac{F_h}{D_o} \right)^{-0.083}$$

$D_h$  : Hydraulic diameter ( $D_o + 2 * F_h$ )  
 $F_p$  : Fin pitch  
 $F_{th}$  : Fin thickness  
 $F_h$  : Fin height  
 $D_o$  : Tube outer diameter without fin  
 $Re = \frac{rV_{max} D_{ho}}{m}$

The air-side area is computed as the total fin and tube area. With this heat transfer coefficient, the effective air-side area is calculated using equation (1.15).

$$A_{eff} = (1 - A_w / (A_t + A_w) * (1 - h_{fin})) * (A_t + A_w) \quad (1.15)$$

The subscripts t and w indicate tube and fin areas respectively. The fin efficiency is calculated using radial fin efficiency expression developed by Kraus et al. [29].

The simulations were performed for manufacturer's specified refrigerant inlet conditions and air-flow rates. The results for the different condensers are shown in Tables 1.6 thru 1.8

Table 1.6 Spiral condenser simulation results (spiral#1)

Parameters	Results							
	Computed				Experimental			
Air flow rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref. Side)	114	141	168	170	106.6/ 109.1	138.1/ 135.9	163.4/ 163.8	167.5/ 166.3
$\Delta P_{ref}$ (kPa)	2.80	2.50	2.10	1.50	*	*	*	*
Exit quality	0.32	0.12	-	-	0.36	0.16	-	-
$T_{ref,cond,out}$ (°C)	40.4	40.4	32.1	30.1	40.2	40.1	35.6	33.8
$T_{air,cond,out}$ (°C)	38.7	37.5	36	34.6	*	*	*	*

- sub-cooled condenser exit

\* data not available

Table 1.7 Spiral condenser simulation results (spiral#2)

Parameters	Results							
	Computed				Experimental			
Air flow rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref. Side)	107	129	156	169	90.9/ 95.6	116.1/ 115	146.7/ 144.3	163.8/ 164.2
$\Delta P_{ref}$ (kPa)	2.60	2.10	1.80	1.40	*	*	*	*
Exit quality	0.39	0.21	0.01	-	0.46	0.32	0.10	-
$T_{ref,cond,out}$ (°C)	40.4	40.4	40.4	30.7	40.1	40.1	39.6	35.4
$T_{air,cond,out}$ (°C)	38.2	36.8	35.6	34.6	*	*	*	*

- sub-cooled condenser exit

\* data not available

Table 1.8 Spiral condenser simulation results (spiral#3)

Parameters	Results							
	Computed				Experimental			
Air flow rate (m <sup>3</sup> /min)	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref. Side)	92	112	136	154	90.4/ 91.8	112.9/ 110.9	141.6/ 138.4	161.5/ 161.5
$\Delta P_{ref}$ (kPa)	2.50	2.10	2.0	1.60	*	*	*	*
Exit quality	0.49	0.34	0.16	0.03	0.49	0.35	0.15	-
$T_{ref,cond,out}$ (°C)	40.4	40.4	40.4	40.4	40.3	40.1	39.9	37.6
$T_{air,cond,out}$ (°C)	37	36	34.9	34.2	*	*	*	*

- sub-cooled condenser exit

\* data not available

### 1.3.2 Sawtooth condenser simulation

The geometry of sawtooth condenser and airflow arrangement are shown in Fig. 1.3. Manufacturer's condensers did not have any clearance; hence, condenser amplitude was equal to the duct height.

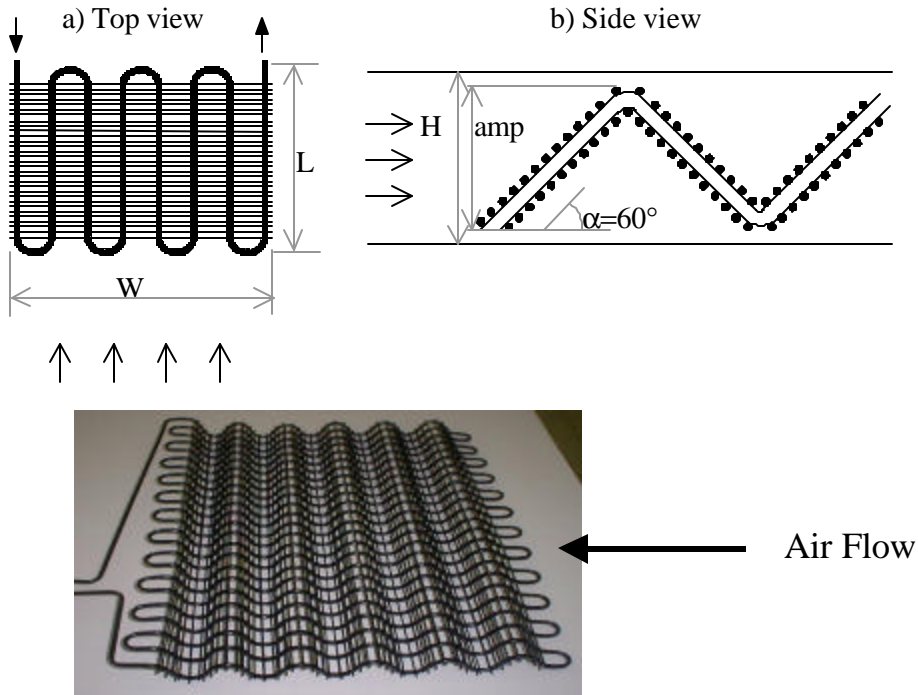
For analysis of this heat exchanger, two zones were considered: two-phase and superheated. Each zone was further subdivided into 4 elements (3 elements having one row of holes at the bottom and 4<sup>th</sup> element having no holes). The data supplied for the ratio of airflow from the bottom to the airflow from the front (7/3) was used in this analysis. Each element was modeled using effectiveness-NTU method as cross-flow heat exchanger with both fluids unmixed. For modeling single-phase zones of such condensers, Barnes and Bullard [30] have used the assumption of parallel-counterflow. However, in these simulations, detailed representation of airflow patterns was needed, so approximations were made for modeling the refrigerant side. The refrigerant-side heat transfer coefficient in both single and two phase were found out using the correlations described in earlier section 1.1 Simulation of vertical cross-counterflow condenser. In the open literature, only Petroski and Clausing [2] have developed correlations for airside heat transfer coefficient and pressure drop across the condenser. Accordingly, the air-side heat transfer coefficient was obtained from the following Nusselt number definition.

$$Nu_w = 0.112 Re_{\max}^{0.667} \quad (1.16)$$

The air-side pressure drop (across the condenser alone) can be found using equation (1.17).

$$\Delta P_{a,c} = C_D \left( \frac{1}{2} \rho V_{\max}^2 \right)$$

$$C_D = 72.7 Re_{\max}^{-0.603} \quad (1.17)$$



c) Photo of sawtooth condenser

Fig. 1.3 Sawtooth condenser configuration and airflow arrangement

The definitions for maximum velocity and Reynolds number are similar to those in equations (1.11) above. Petroski and Clausing [2] has obtained these correlations based on sawtooth condensers having 7-9 layers. The number of layers in condenser prototypes simulated here were larger than these, hence, corrections were applied to the pressure drop across coil as given in equation (1.18).

$$N_{layers} = \frac{2 * Depth}{\left( \frac{Height}{\sin(60)} \right)}$$

$$\Delta P_{corrected} = \Delta P_{a,c} * \frac{N_{layers}}{7} \quad (1.18)$$

The number of layers is related to the depth and height of the condenser through the sawtooth angle ( $60^\circ$ ) from the data supplied by the manufacturer [19]. The total pressure drop across the condenser is the weighted sum (over the condenser depth) of the condenser pressure drop across each element. The duct pressure drop is computed corresponding to the duct depth for each element. Again, the pressure drop across the front grille is neglected in these computations. The results of these simulations for condensers S/T#1 and S/T#2 are presented in Tables 1.9 & 1.10 respectively. One interesting observation is that the experimental measurements of air-side pressure drop are lower than those predicted by the simulation model. This is unexpected because, the measurements included pressure drop across the front grille and the pressure drop experienced by air due to turning at the back plate to converge to the hole for the fan.

Table 1.9 Sawtooth condenser simulation results (S/T#1)

Parameters	Results							
	Computed				Experimental			
Air flow rate ( $\text{m}^3/\text{min}$ )	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref-side)	90.1	116	149	179.	99.7/ 101.6	128/ 123.6	160.4/ 161.3	170.9/ 167.6
$\Delta P_{\text{air}}$ (mmH <sub>2</sub> O)	0.78	1.3	2.25	3.34	0.34	0.67	1.48	2.56
$\Delta P_{\text{ref}}$ (kPa)	6.5	6.4	6.4	6.4	22	26	19.6	12
Exit quality	0.50	0.31	0.07	-	0.42	0.26	-	-
$T_{\text{ref,cond,out}}$ ( $^\circ\text{C}$ )	40.2	40.3	40.3	40.3	39.7	39.6	38.4	32.3
$T_{\text{air,cond,out}}$ ( $^\circ\text{C}$ )	37	36.2	35.4	34.7	36.8	36.2	35.3	34.2

- sub-cooled condenser exit

Table 1.10 Sawtooth condenser simulation results (S/T#2)

Parameters	Results							
	Computed				Experimental			
Air flow rate ( $\text{m}^3/\text{min}$ )	0.67	0.97	1.43	1.90	0.67	0.97	1.43	1.90
Capacity (W) (air/ref-side)	106	136	177	213	106/ 108	138/ 136	166/ 167	167/ 169
$\Delta P_{\text{air}}$ (mmH <sub>2</sub> O)	0.88	1.5	2.5	3.76	0.4	1.0	1.6	2.8
$\Delta P_{\text{ref}}$ (kPa)	6.5	6.4	6.4	6.4	21	26	14	7
Exit quality	0.39	0.16	-	-	0.37	0.17	-	-
$T_{\text{ref,cond,out}}$ ( $^\circ\text{C}$ )	40.3	40.3	40.3	40.3	39.7	39.6	38.4	32.3
$T_{\text{air,cond,out}}$ ( $^\circ\text{C}$ )	38.1	37.2	36.3	36	37.3	36.7	35.5	34.2

- sub-cooled condenser exit

#### 1.4 Air-side pressure drop assumptions

The experimental results on the air-side pressure drop of the manufacturer's condensers include in addition to duct and condenser pressure losses, the pressure drop across the front grille and the pressure drop experienced by air due to turning to converge at the hole where the fan is located. In their domestic refrigerators, in addition to these losses the fan has to perform extra work to pull the air across the compressor also. The results of the simulation do not include these additional components of air-side pressure drop. In this section, the air-side pressure

drop due to front grille is computed for the air-flow rates considered by the manufacturer in their experiments. The manufacturer's front grille is (40 mm x 484 mm). It has 45 holes (length = 20.0 mm, breadth = 5 mm and 2.5 mm radii on both ends). Only 30% of the total air flows across this front grille. Based on this, the air velocity through each hole is first computed. The holes are treated as orifices in a plate. Reynolds number was evaluated based on the inlet flow rate. For laminar air flow, the following formula for  $C_d$  was used for computations of pressure drop from Idelchik [31].

$$C_d = [z_j + e_D^{-Re} (1.707 - f^2)] / f^2$$

$$z_j = a2 + \frac{b2}{Re} + \frac{c2}{Re^2} + \frac{d2}{Re^3}$$

$$e_D^{-Re} = a1 + b1 * Re + c1 * Re^2 \quad (1.19)$$

$f$  = area of holes/area of front plate. The seven parameters in the above curve fits of functions of  $Re$  were obtained by doing a least squares fit of these parameter values given in Idelchik [31] over the Reynolds number range of interest (60 thru 1000). The grille pressure drop was computed as follows.

$$\Delta P_{grille} = 1/2 * C_d * \rho * V^2 \quad (1.20)$$

The values obtained for various air-flow rates have been listed in table 1.11 The exit pressure drop is computed from the following expression of pressure drop for (air flow from conduit of one size to another, with  $Re < 10^5$ ) from [31].

The relevant equations are:

$$C_{d,exit} = z_j + e_0^{-Re} * (z_0 - F_0 / F_2)^2$$

$$z_0 = 1 + 0.707 * \sqrt{1 - F_0 / F_1} \quad (1.21)$$

where,

$F_0$  = hole area

$F_{1/2}$  = cross-sectional area of conduit from where flow is (leaving/entering)

In these computations, hole area is assumed to be equal to the area of conduit where air is entering. (i.e.  $F_0 = F_2$ ). In [31], tables are available for computing the parameters  $\zeta_\phi$  and also for  $(\epsilon_0)^{-Re}$  as a function of Reynolds number and  $F_0/F_1$ . Substituting the values for different flow rates and ( $F_0/F_1 = 121 \times 40 / 484 \times 40 = .25$ ), the exit pressure drop has been calculated.

Table 1.11 Additional pressure drop

Air Flow Rate (m <sup>3</sup> /min)	$\Delta P_{grille}$ (mmH <sub>2</sub> O)	$\Delta P_{exit}$ (mmH <sub>2</sub> O)
0.67	0.4	0.12
0.97	0.8	0.26
1.43	1.6	0.55
1.9	2.5	0.97

The air flow rates listed in this Table are the manufacturer’s test conditions. Actually only 30% of this enters through the front grille. For exit pressure drop computations, velocity is computed based on the total air flow rates. From Table 1.11, it is seen that this additional pressure drop is about 60% larger (for C/F#1 condenser for the lowest air-flow rate) than the total of condenser and duct loss.

**1.5 Condenser performance in actual refrigerator**

The foregoing discussion is based entirely on simulations in wind tunnel. The test conditions in wind tunnel experiments differ significantly from the actual refrigerator. Therefore, to account for difference in performance of condensers in wind tunnel and actual refrigerator, additional experiments were performed in a simulated machine room to replicate the fluid mechanics of the air flow beneath an actual refrigerator. In the machine room, the air exiting from the fan has to flow over the compressor and through an exit grille. This additional pressure drop increases fan power and decreases air flow rate. This, in turn, decreases the condenser capacity. The machine room experimental results for the ET-basic and S/T#2 condensers (see Table 1.1 for dimensions) are shown in Table 1.12.

Table 1.12 Machine room experimental results

Condenser Configuration	Box-type			Saw-tooth		
	2.5	3.0	3.5	2.5	3.0	3.5
Refrigerant flow rate (kg/h)	2.5	3.0	3.5	2.5	3.0	3.5
Inlet pressure (kPa)	1033	1034	1033	1033	1034	1034
Inlet temperature (°C)	65	65	65	65	65	65
Exit quality	-	0.17	0.30	-	-	0.23
Capacity (W)	137.2	135.8	138.0	138.7	160.5	148.4

Box type condenser ET-basic condenser in Table 1.1 Sawtooth type condenser: S/T#2 condenser in table 1.1

**2. Optimization of sawtooth condenser**

Optimization of sawtooth condenser was conducted using the same equations as the simulation model developed in section 1.3. Detailed descriptions of the optimization are given in this section of the report.

**2.1 Simulation of wind tunnel experiments**

The dimensions of two prototype sawtooth heat exchangers are described in Table 1.1. Inlet air and refrigerant conditions are specified in Table 1.2. The simulation results are compared to the manufacturer’s experimental results from wind tunnel tests earlier in this report. Since the refrigerant-side capacity calculations are all based on measurements obtained with immersion RTD’s and a mass flow rate maintained at 3.0 kg/h for all the tests, they are expected to be more consistent than calculations based on air side measurements. At (air-flow rate = 1.43 m<sup>3</sup>/min) 168 W (from [19]), heat rejection from the baseline condenser in the wind tunnel is greater than prototype ST#1 (149 W) and slightly less than ST#2 (177 W) as seen from the simulation results in Tables 1.9-10.

According to the wind tunnel tests, the new prototype condensers fail to provide substantial performance improvement over that of the baseline unit. However this is not the case in practice. When installed in a refrigerator, the baseline condenser is not well ducted, because flow configuration is not uniform due to the defrost water pan located at the bottom of condenser and oblique air entrance to the condenser. The inlet air enters the condenser from the backside of refrigerator and flows perpendicular to entrance direction through the condenser and

compressor and exits to the backside. Thus performance is degraded due to air bypass effects, and the resulting refrigerant exit quality is greater than zero. Downstream, a hot wall condenser (cluster) is therefore needed to complete the condensation process. Because of the superior ducting in the wind tunnel, however, the baseline condenser rejects enough heat to subcool the outlet about 6°C. The prototype condensers, on the other hand, are designed to fit in a duct under the front of the refrigerator, allowing the machine compartment to be downsized to enlarge the food storage volume. Since the sawtooth prototypes would be installed in a duct similar to the wind tunnel experiment, their performance in the actual refrigerator is expected to be the same as in the wind tunnel.

## 2.2 Simulation of compressor-condenser subsystem

The wind tunnel experiments were conducted with identical inlet conditions [19], but the condenser outlets were subcooled. Due to the subcooling differences, it is difficult to estimate potential contributions to cycle efficiency. Although it is difficult to be certain without simulating the entire system, having a subcooled condenser outlet is probably suboptimal – saturated liquid is probably a more reasonable design target because it maximizes the two-phase area (and minimizes charge, which in turn minimizes cycling losses associated with charge migration). Therefore a second set of simulations was conducted to compare the three condensers (baseline, ST#1 and ST#2), with the compressor included within the subsystem boundary so the differences in compressor work could be compared.

The compressor model was developed from manufacturer's data as discussed in earlier sections of this report. For this analysis it is assumed that the refrigerator is operating at the 30°C design condition where the compressor inlet state, obtained from the manufacturer, is  $P = P_{\text{sat}}(T_e)$  where  $T_e = -35^\circ\text{C}$ , and  $T_{\text{suc}} \sim 30^\circ\text{C}$ ) and the evaporator exit is saturated vapor. The baseline cycle was simulated in the following manner, with the result shown in Figure 2.1. Based on the manufacturer's data for the baseline coil for air-flow rate = 1.43 m<sup>3</sup>/min, an approximate value for the air-side heat transfer coefficient was extracted using a simple multi-zone model of the condenser. This was then used to simulate the baseline cycle with specified compressor inlet condition with the condenser exit fixed to be saturated liquid, yielding the compressor outlet pressure. The fan power was assumed to be constant at 2.16 W. The various parameters are reported in Table 2.1, and results of a simple thermodynamic state point analysis are shown in the P-h diagram. The total refrigerant pressure drop was assumed to be 1 C.



Table 2.1 Prototype condenser simulation

Parameters	Results		
	Baseline	S/T#1	S/T#2
Air Flow Rate (m <sup>3</sup> /min)	1.43	1.43	1.43
Condenser Capacity (W)	157.7	153	158
Evaporator capacity (W)	146	142	147
Fan power (W)	2.16	2.16	2.16
Compressor power (W)	125.6	125	125.2
COP	1.144	1.121	1.155
Refrigerant Mass Flow Rate (kg/h)	3.0	2.95	3.0
T <sub>dis</sub> (°C)	62.5	63.2	62.9
T <sub>cond</sub> (°C)	39.3	40.8	39.2

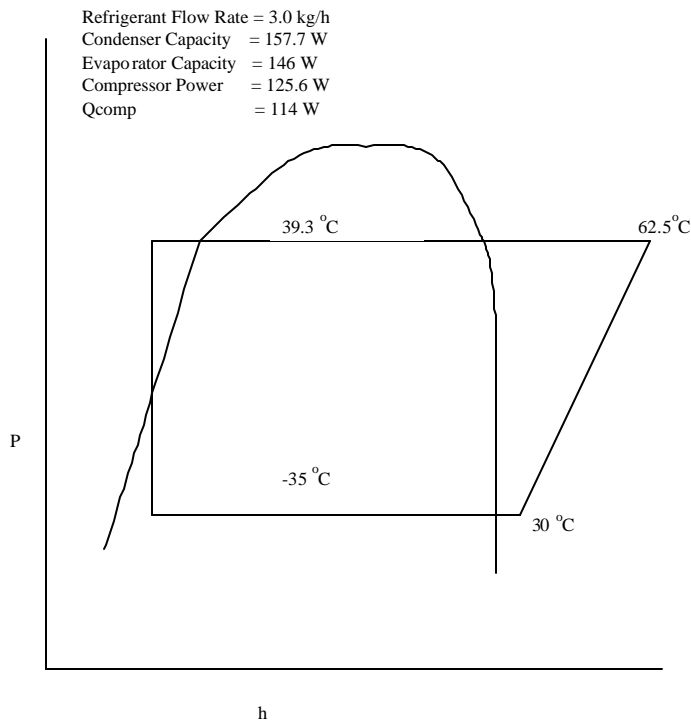


Fig. 2.1 Refrigeration Cycle on P-h diagram

The results show that the ST#2 design offers negligible performance improvement over the baseline design. This was not surprising because the wind tunnel experiments [19] showed that the baseline and S/T#2 coils exhibited approximately the same heat transfer capacities. Recall that the baseline coil was positioned in the wind tunnel such that no air bypassed the coil, hence its performance was better than in the actual refrigerator. Upon closer inspection, however, there are reasons to expect that the higher heat transfer coefficient of the sawtooth design should have improved performance, even in the wind tunnel. The heat transfer areas of these two coils are

same (Table 1.1). From the wind tunnel data, the air side heat transfer coefficient was found to be  $55 \text{ W/m}^2\text{-K}$ , compared to  $69 \text{ W/m}^2\text{-K}$  for the ST#2 heat exchanger. The explanation is clear. The higher  $h$  of the sawtooth coil was exploited to produce a packaging advantage: allowing 70% of the air to enter the duct from the bottom, so it passed over only part of the heat exchanger. That explains why the overall  $Q$ 's were the same, despite the large difference in air side heat transfer coefficient. There were also slight differences between the two heat exchangers on the refrigerant side: tube length was ~20% greater for the baseline coil (Table 1.1). However, the teeth in the saw-tooth coil produce a higher refrigerant-side pressure drop (associated with turning of the refrigerant and perhaps also due to gravity).

These simulations and wind tunnel results both illustrate how the prototype ST#2 outperforms ST#1. The slightly higher refrigeration capacity provided by ST#2 means that the refrigerator should pull down more quickly. It should also have slightly shorter runtimes than the baseline, even if perfect ducting of the baseline unit enabled it to perform as well in a refrigerator as it did in the ducted wind tunnel test. The results are directly comparable, except that the sawtooth simulations account for refrigerant-side pressure drop while the baseline calculation neglects it (a small effect, affecting condenser heat transfer by only about 1Watt). The additional benefit of the sawtooth design, of course, is that it allows for improved packaging in the machine room, and therefore enlargement of the refrigerated space.

If a hot-wall condenser (cluster) were located downstream of the sub-condenser, it would have little effect on the cycle at this design operating condition, because the additional ( $\sim 3^\circ\text{C}$ ) subcooling in the cluster would increase (by only  $\sim 2^\circ\text{C}$ ) the amount of subcooling occurring in the ctslhx (capillary tube suction line heat exchanger), thus having only a small effect on evaporator inlet quality and hence refrigerant mass flow rate and power. Therefore the foregoing simulations and the optimization analysis that follow are conservative, because they neglect the presence of the cluster.

### 2.3 Optimization of sawtooth geometry

The next step was to conduct an optimization analysis to find the best wire and tube diameters and spacings for a sawtooth condenser designed to fit in the same package volume as the prototype units. The constraints on optimization specified by the manufacturer are listed in table 2.2 below, along with slightly relaxed constraints used by Barnes and Bullard [30] and the wire and tube sizes and spacings used by Petroski and Clausing [2] when developing the air-side heat transfer and pressure drop correlations. The constraints listed under the column named search are the manufacturer's constraints [19]; and the constraints listed under the columns named Barnes and Correlation are the constraints in [30] and [2] respectively. Constraints on wire diameter were not specified by manufacturer, so the upper bound was arbitrarily set to 1.8 mm.

Table 2.2 Constraints on optimization

Lower Bound			Variable	Upper Bound		
Barnes	Search	Correlation		Correlation	Search	Barnes
-	15	25.4	Tube Pitch (mm)	25.4	20	-
4	4.2	4.8	Tube Diameter (mm)	4.8	4.76	7.95
1	7	4.8	Wire Pitch (mm)	6.4	10	10
0.9	0.9	1.22	Wire Diameter (mm)	1.6	1.8	1.8
-	-	-	Tube –wall –thickness/diameter ratio=70/476	-	-	-
4	-	-	Tube –pitch/ diameter-ratio	-	-	-

The same refrigerant and air inlet conditions are specified for the compressor/condenser subsystem boundary:  $P_{suc}$ ,  $T_{suc}$ ,  $T_{amb}$ , and the refrigerant outlet quality is fixed at  $x=0$ . Instead of specifying wire and tube diameters and spacings, the optimization searches for the combination that maximizes system

$$COP = \frac{Q_e}{\dot{W}_{fan} + \dot{W}_{comp}} \quad (2.1)$$

From the fan data provided by the manufacturer [19], a quadratic least squares curve fit was used to obtain fan curve. Combined fan and motor efficiency was set to 12% (by using manufacturer’s data for fan power at air-flow rate = 1.43 m<sup>3</sup>/min). Instead of specifying air flow rate, the optimal value is calculated from the fan curve, from the air side pressure drop that varies with condenser geometry. The maximum COP was achieved when the addition of heat transfer surface area reduced compressor power by the same amount as the fan power increased.

The optimal geometry is shown in column 1 of Table 2.3. The second column shows the effect of limiting the search to the range of wire and tube dimensions that provided the experimental basis for developing the correlation, thus avoiding extrapolation. The third column allows for significant extrapolation, expanding the search domain over a broader range of dimensions suggested by manufacturers of wires and tubes (see Barnes & Bullard [30]).

As expected, the highest COP was obtained by expanding the search domain to that specified by Barnes [30]. Given the packaging constraints, it appears that performance is optimized by using smaller diameter tubes and spacing them closer together, with large diameter wires spaced more closely than ST#2. The closely spaced wires would increase air side pressure drop, so the air flow rate was reduced to minimize the fan power penalty. As a result, the outlet air temperature increased to the point where it almost equals the condensing temperature (see column 3 of table 2.3). Any lower air flow rate would have required a higher condensing temperature, thereby decreasing COP. Basically, the optimization algorithm sought to improve the condenser by adding surface area, and by reducing air flow rate to the minimum. Whether search constraints or Barnes constraints [30] are used, the COP is about the same, thus, the configuration listed in column 1 of the table 2.3 is optimal under these circumstances.

Table 2.3 Results of optimization (constant fan efficiency)

Parameter	Max COP (Search Constraints)	Max COP (Correlation constraints)	Max COP (Barnes' Constraints)	ST#2
COP	1.136	1.12	1.139	1.11
Evaporator Capacity (W)	145	142	145.	142
Total Power (W)	127.6	127.4	127.5	127.3
Compressor Power (W)	125	125	125	124.8
Fan Power (W)	2.55	2.54	2.5	2.5
Condenser Capacity (W)	155.3	152	156	151.8
Air Flow Rate (m <sup>3</sup> /min)	1.06	1.04	0.9	1.08
Tube Diameter (mm)	4.2 <sup>(-)</sup>	4.8 <sup>(+/-)</sup>	4 <sup>(-)</sup>	4.76
Wire Diameter (mm)	1.8 <sup>(+)</sup>	1.6 <sup>(+)</sup>	1.8 <sup>(+)</sup>	1.6
Tube Spacing (mm)	15 <sup>(-)</sup>	25.4 <sup>(+/-)</sup>	16	20
Wire Spacing (mm)	7 <sup>(-)</sup>	4.8 <sup>(-)</sup>	4	7
T <sub>cond</sub> (°C)	39.8	40.9	39.8	41
T <sub>air,out</sub> (°C)	37.5	37.5	38.9	37.2
A <sub>cond</sub> (m <sup>2</sup> )	0.534	0.495	0.62	0.47
Mass (kg)	2.	1.75	2.	1.72
D P <sub>ref</sub> (kPa)	15.5	4.7	18.5	6.1
D P <sub>air</sub> (mmH <sub>2</sub> O)	1.8	1.8	1.93	1.73
T <sub>dis</sub> (°C)	63.1	63.2	63.1	63.2

(-)=> value at lower bound

(+)=> value at upper bound

#### 2.4 Effect of constant fan power

For all the optimizations described in section 2.3 above, the compressor power was computed using the compressor map for the Samsung ZK180b. The combined fan-motor efficiency was assumed constant at 12% and fan power was calculated for each combination of airflow and pressure drop.

At this point the authors began to question an assumption embedded in the model, namely that the compressor fan/motor efficiency was constant. Data provided later by the manufacturer showed significant variations in fan efficiency (manufacturer's data [19] indicated that the fan power is fairly constant over a wide range of air-flow rates and air-side pressure drops, hence, it was concluded that fan efficiency has to vary), so the analysis was repeated using the constant power data (~2.16 W, for most of air-flow rates in the range of interest) provided by manufacturer [19]. These data show that fan power is essentially independent of pressure drop, over the relevant range of air flow rates. The revised results are shown in table 2.3a below. The authors expected further addition of area when fan power penalty is kept constant. The results in table 2.3a, however, indicate that further addition of area is precluded by the requirement of certain air-flow rate to condense the refrigerant. Hence, these results are very similar to the ones listed in table 2.3, the slight improvement in COP is observed only because of reduction in fan power penalty.

Table 2.3a Results of optimization (constant fan power)

Parameter	Max COP (Search Constraints)	Max COP (Barnes' Constraints)
COP	1.14	1.141
Evaporator Capacity (W)	145	145.1
Total Power (W)	127.2	127.2
Compressor Power (W)	125	125
Fan Power (W)	2.16	2.16
Condenser Capacity (W)	155.3	156
Air Flow Rate (m <sup>3</sup> /min)	1.06	0.9
Tube Diameter (mm)	4.2 <sup>(-)</sup>	4 <sup>(-)</sup>
Wire Diameter (mm)	1.8 <sup>(+)</sup>	1.8 <sup>(+)</sup>
Tube Spacing (mm)	15 <sup>(-)</sup>	16
Wire Spacing (mm)	7 <sup>(-)</sup>	4
T <sub>cond</sub> (°C)	39.8	39.8
T <sub>air,out</sub> (°C)	37.5	38.9
A <sub>cond</sub> (m <sup>2</sup> )	0.534	0.62
Mass (kg)	2	2.24
D P <sub>ref</sub> (kPa)	15.5	18.4
D P <sub>air</sub> (mmH <sub>2</sub> O)	1.77	2
T <sub>dis</sub> (°C)	63.1	63.1

(-)=> value at lower bound

(+)=> value at upper bound

### 2.5 Optimization of sawtooth condenser in cross-counterflow arrangement

Based on earlier work conducted at ACRC, (see [2], [27] and [30]) it was expected that the vertical cross-counterflow condenser would give the best performance. For manufacturing reasons, however, the sawtooth condenser is desirable because it is possible to exploit the higher air-side heat transfer coefficient at lower duct heights, thus leading to an overall increase in the refrigerator compartment volume. The results of optimization of sawtooth condenser when air-flow is parallel to tubes are presented in Sections 2.3-4. It was decided to explore the sawtooth condenser by placing it in air-flow configuration similar to the vertical cross-counterflow arrangement. The simulation model, described in an earlier section 2.1 was used for this purpose. The only changes made were to the air-side heat transfer and pressure drop correlations [2]. The geometry calculations, too, were different. The same general dimensions of the duct were used. Optimization was carried along similar lines as for optimization of the sawtooth condenser.

A four dimensional search using EES [18] is inherently complex task. This is further complicated by the finite element of nature of the algorithm used. Hence, from insights gained during earlier analyses, the optimal configuration is expected to be the one that maximizes the heat transfer area. Hence the tube spacing was fixed at the lowest bound and holding tube diameter constant at its upper bound, a two-dimensional search was carried out to determine the optimal configuration. This led to the configuration with the lowest number of thickest wires. Varying the diameter in the range specified by manufacturer's constraints had no further effect on the value of COP.

The results of this optimization are presented in table 2.4 below. Since the heat transfer area is already at its maximum, the optimization with fan power constant is unlikely to yield any further performance improvement.

This configuration seems to be only slightly better than the optimal configuration found out using sawtooth condenser in cross-parallel flow configuration.

Table 2.4 Optimization of cross-counterflow sawtooth condenser

Parameter	Max COP Configuration (manufacturer's constraints)
COP	1.163
Evaporator Capacity (W)	149.3
Total Power (W)	128.4
Compressor Power (W)	125.8
Fan Power (W)	2.6
Condenser Capacity (W)	161.3
Air Flow Rate (m <sup>3</sup> /min)	1.08
Tube Diameter (mm)	4.76 <sup>(+)</sup>
Wire Diameter (mm)	1.8 <sup>(+)</sup>
Tube Spacing (mm)	15 <sup>(-)</sup>
Wire Spacing (mm)	7 <sup>(-)</sup>
T <sub>cond</sub> (°C)	38.5- 38.0
T <sub>air,out</sub> (°C)	37.7
A <sub>cond</sub> (m <sup>2</sup> )	0.84
Mass (kg)	3.5
D P <sub>ref</sub> (kPa)	15.0
D P <sub>air</sub> (mmH <sub>2</sub> O)	1.74
T <sub>dis</sub> (°C)	62.2

(-)=> value at lower bound

(+)=> value at upper bound

## 2.6 Discussion of results

The foregoing analysis was based on compressor performance data obtained at condensing temperatures ranging from 40 to 48°C. The data points are shown in Fig. 2.2, along with curve fits produced with the standard 10-parameter polynomial equation. Note that the curve fit must be extrapolated to condensing temperatures less than 40°C, and that the extrapolated curve appears to be physically unrealistic. This illustrates the risks of extrapolating polynomials; they can fit very well within the range of actual data, but they embody none of the physics governing compressor behavior.

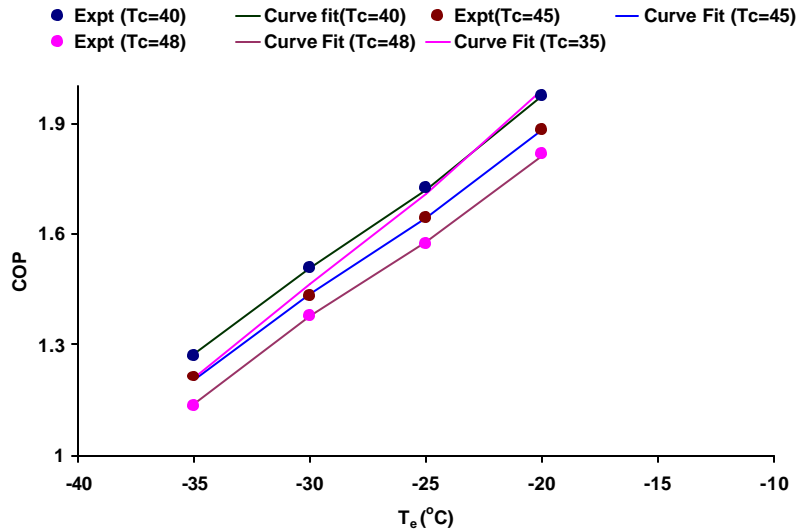


Fig. 2.2 COP vs Te at constant Tcond (polynomial fit)

Fig. 2.3 shows the curve fit obtained by Kim and Bullard's semi-empirical model [3], which is essentially a curve fit of physical parameters known to govern compressor operation. It does not fit the data quite as well inside the range, but it appears to extrapolate to 35°C in a much more reasonable manner. Therefore Kim and Bullard's compressor model [3] was used for all the simulations and optimizations presented in this report. Fig. 2.3 was obtained by calculating  $COP_{comp}$  from the original measurements of refrigeration capacity and power in the compressor calorimeter. At -35°C evaporating temperature, power was essentially independent of condensing temperature; it changed by less than 1% over the 40-48°C range of experimental data. The system COP improvements shown in Tables 8 and 9, therefore, resulted from the capacity increase that was experienced as condensing temperature was reduced. From the P-h diagram it can be seen that lower condensing temperatures produce a more efficient cycle, by reducing compressor power per unit mass, and at the same time increasing the refrigerating effect per unit mass by reducing evaporator inlet quality. Referring again to the optimization results, it is clear that the condenser geometries improved system COP by improving the cycle (and therefore refrigerating capacity), not by directly reducing compressor power.

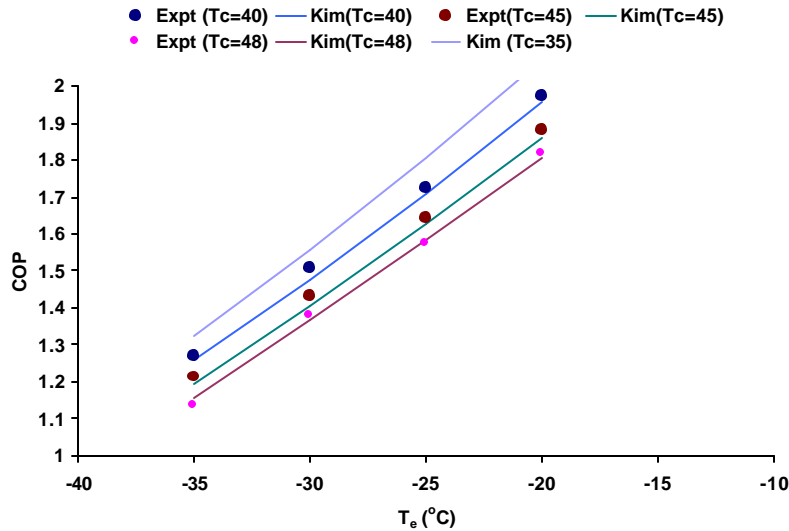


Fig. 2.3 COP vs.  $T_e$  at constant  $T_{cond}$  (Kim and Bullard [3])

Referring again to 35°C and 40°C curves in Fig. 2.3 it is apparent that condenser improvements can increase  $COP_{comp}$  at any evaporating temperature. Moreover the percentage improvement rather small, and is nearly independent of evaporating temperature. In contrast, substantial  $COP_{comp}$  increases could be obtained by improving evaporator performance, raising  $T_e$  from -35 to -25°C for example.

In summary, the simulations have illustrated how a sawtooth condenser could increase system performance, given constraints on package volume and fan performance. System performance improvements resulted mainly from reducing the condensing temperature – the compressor power remained almost constant while refrigerant mass flow and system capacity increased, as heat transfer area was added to the condenser in a manner that minimized air and refrigerant side pressure losses. The optimal design was reached when the added condenser area caused so much pressure drop that the air flow rate dropped to the point where the air and refrigerant temperatures pinched. As the temperature difference diminished, condenser mass, a surrogate for cost, began to rise rapidly. The conclusion is that using a more powerful and efficient fan might be the most cost-effective way to achieve further increases in condenser performance. That would allow air flow rate to be increased and the condensing temperature reduced, and leave a slightly larger  $dT$  at the pinch point, assuming that the power requirement and noise generation would not be excessive. It appears that noise and cost constraints will severely limit the potential for achieving greater performance from wire-on-tube condenser design.

The primary benefit of the sawtooth condenser, compared to the baseline design, appears to come from packaging rather than changes in heat transfer coefficient and surface area. The package volume is not only ducted, but also allows for food storage volume to be increased.

Greater benefits, however, are likely to be obtained by efforts to improve evaporator performance. At higher evaporating temperatures, substantial compressor power savings could be obtained. Moreover, warmer fin surfaces collect less frost and therefore maintain better food quality and minimize defrost energy.



The optimal condenser configuration prescribed here was obtained using fan curve and air-side pressure drop across duct and condenser alone. This analysis neglected the pressure drop across the front-grille, the turning experienced by air to converge at the hole where fan is located, and pressure drops across the compressor and the exit grille. As illustrated in section 1.4, these pressure drops could be major components of total pressure drop on the air-side. However, certain trends are apparent from the optimizations presented in this chapter, and it seems reasonable to assume that the same geometries and air flow rates would also be optimal if a slightly more powerful fan were used to overcome the additional air side pressure drop. The prescribed condenser outlet condition (saturated liquid) determines the condensing temperature. Then according to fan characteristic curve, the lowest air-flow rate is calculated to avoid pinching. If air-side pressure drop rises substantially due to the downstream obstructions, the optimal condenser may be more open (i.e. the condenser will have lower heat transfer area), shifting the power requirement to the compressor and resulting in a slightly higher condensing temperature.

It is recommended that future experimental work focus on perfecting the “machine room experimental technique” being developed by the manufacturer. With additional instrumentation to measure air flow rates, for example, performance can be measured under realistic conditions. By including the compressor and exit grille, the fan can be properly sized to achieve the desired air flow rate. By controlling the compressor suction state and using a needle valve to control the condenser outlet at the saturated liquid state, the subsystem will adjust to the condensing temperature that will be experienced in the actual refrigerator. Such an approach ensures optimal matching of the condenser and compressor subsystem. The resulting mass flow rate determines the refrigerating capacity, and hence the runtime fraction for the compressor chosen. The simulation and optimization modeling presented here can help identify design strategies and near-optimal condenser configurations prior to conducting such experiments.

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## Appendix A. Data sets used for the model and simulation results

A1 Data set I

$T_{evap}$ (°C)	$T_{cond}$ (°C)	$T_{suc}$ (°C)	$T_{shell}$ (°C)	$T_{amb}$ (°C)	$T_{dis,exp}$ (°C)	$T_{dis,cal}$ (°C)	$m_{r,exp}$ (kg/h)	$m_{r,cal}$ (kg/h)	$W_{exp}$ (W)	$W_{cal}$ (W)	$h_c$	$h_v$
-35	55	32.2	60.1	32.2	62.6	-	1.65	-	101	-	-	
-35	48	32.2	59.7	32.2	63.3	62.2	2.01	2.13	106	105	0.51	0.64
-35	45	32.2	59.7	32.2	63.5	62.2	2.18	2.22	108	107	0.53	0.67
-35	40	32.2	59.4	32.2	63.5	61.8	2.34	2.36	108	108	0.54	0.71
-30	55	32.2	63.2	32.2	63.2	74.5	2.82	2.85	127	126	0.57	0.66
-30	48	32.2	62.8	32.2	71.6	73.4	3.1	3.09	129	129	0.58	0.72
-30	45	32.2	62.4	32.2	71.2	72.6	3.2	3.19	128	129	0.59	0.74
-30	40	32.2	61.5	32.2	71.7	70.9	3.34	3.33	128	129	0.58	0.77
-25	55	32.2	66.2	32.2	76.6	83.7	4.01	4.06	153	153	0.60	0.73
-25	48	32.2	65.3	32.2	79.3	80.9	4.37	4.32	153	153	0.61	0.77
-25	45	32.2	64.6	32.2	79.3	79.5	4.51	4.42	152	153	0.61	0.79
-25	40	32.2	63.4	32.2	78.7	76.9	4.71	4.57	150	151	0.61	0.82
-20	55	32.2	68.3	32.2	86.3	89.2	5.54	5.60	181	181	0.62	0.78
-20	48	32.2	66.9	32.2	85.0	85.3	5.88	5.87	178	179	0.62	0.82
-20	45	32.2	65.4	32.2	84.3	83.5	6.01	5.97	176	177	0.61	0.83
-20	40	32.2	60.9	32.2	81.0	80.1	6.26	6.14	172	173	0.61	0.86
-15	55	32.2	67.2	32.2	90.4	91.9	7.3	7.55	210	210	0.61	0.82
-15	48	32.2	65.5	32.2	88.6	87.1	7.65	7.84	204	205	0.61	0.86
-15	45	32.2	64.5	32.2	86.8	84.9	7.83	7.95	200	201	0.61	0.87
-15	40	32.2	63.0	32.2	85.0	81.1	8.01	8.12	194	195	0.60	0.89

A2 Data set II

$T_{evap}$ (°C)	$T_{cond}$ (°C)	$T_{suc}$ (°C)	$T_{shell}$ (°C)	$T_{amb}$ (°C)	$T_{dis,exp}$ (°C)	$T_{dis,cal}$ (°C)	$m_{r,exp}$ (kg/h)	$m_{r,cal}$ (kg/h)	$W_{exp}$ (W)	$W_{cal}$ (W)	$h_c$	$h_v$
-35	55	32.2	-	32.2	77.1	81.1	2.67	2.67	142	133	0.50	0.59
-35	48	32.2	-	32.2	77.0	81.5	2.97	2.97	145	139	0.51	0.66
-35	45	32.2	-	32.2	77.8	81.0	3.20	3.09	145	141	0.54	0.68
-35	40	32.2	-	32.2	78.9	79.7	3.27	3.27	145	143	0.52	0.72
-30	55	32.2	-	32.2	85.7	90.3	3.85	3.90	172	167	0.54	0.68
-30	48	32.2	-	32.2	86.7	88.0	4.23	4.21	172	171	0.56	0.73
-30	45	32.2	71.5	32.2	83.4	88.4	4.28	4.34	171	170	0.55	0.75
-30	40	32.2	70.0	32.2	86.2	84.7	4.58	4.52	169	172	0.56	0.78
-25	55	32.2	-	32.2	93.9	95.6	5.28	5.42	206	204	0.56	0.74
-25	48	32.2	71.8	32.2	94.0	91.5	5.69	5.74	202	205	0.57	0.79
-25	45	32.2	70.5	32.2	92.9	90.2	5.94	5.86	200	205	0.59	0.80
-25	40	32.2	68.9	32.2	91.7	87.1	6.15	6.05	197	203	0.58	0.83
-20	55	32.2	74.3	32.2	99.8	98.0	7.09	7.25	239	243	0.59	0.79
-20	48	32.2	72.8	32.2	99.1	92.5	7.58	7.58	234	242	0.59	0.83
-20	45	32.2	71.5	32.2	97.4	90.9	7.72	7.71	231	239	0.59	0.84
-20	40	32.2	69.8	32.2	91.4	90.2	7.91	7.90	224	231	0.58	0.86

A3 Data set III

$T_{evap}$ (°C)	$T_{cond}$ (°C)	$T_{suc}$ (°C)	$T_{shell}$ (°C)	$T_{amb}$ (°C)	$T_{dis,exp}$ (°C)	$T_{dis,cal}$ (°C)	$m_{r,exp}$ (kg/h)	$m_{r,cal}$ (kg/h)	$W_{exp}$ (W)	$W_{cal}$ (W)	$h_c$	$h_v$
-35.0	55.0	32.2	63.4	32.2	66.1	65.7	2.28	2.39	117	114	0.54	0.57
-35.0	48.0	32.2	63.3	32.2	66.9	66.7	2.66	2.68	121	119	0.54	0.64
-35.0	45.0	32.2	63.2	32.2	66.7	66.7	2.86	2.80	122	121	0.54	0.67
-35.0	40.0	32.2	62.6	32.2	70.6	63.7	3.02	2.98	123	124	0.53	0.71
-30.0	55.0	32.2	66.7	32.2	75.7	80.1	3.53	3.54	144	140	0.58	0.66
-30.0	48.0	32.2	66.1	32.2	75.3	78.2	3.90	3.84	146	144	0.58	0.72
-30.0	45.0	32.2	65.1	32.2	77.4	75.3	4.02	3.96	145	146	0.57	0.74
-30.0	40.0	32.2	64.1	32.2	77.5	72.2	4.18	4.15	143	147	0.56	0.77
-25.0	55.0	32.2	68.0	32.2	83.4	89.2	4.96	4.95	173	168	0.61	0.73
-25.0	48.0	32.2	67.5	32.2	83.5	83.9	5.23	5.26	171	172	0.60	0.77
-25.0	45.0	32.2	67.0	32.2	82.9	81.7	5.38	5.39	169	172	0.59	0.79
-25.0	40.0	32.2	65.7	32.2	81.9	77.6	5.55	5.58	166	171	0.58	0.82
-20.0	55.0	32.2	69.3	32.2	88.6	94.6	6.68	6.67	203	198	0.63	0.78
-20.0	48.0	32.2	68.3	32.2	86.7	88.6	6.96	6.99	197	199	0.61	0.82
-20.0	45.0	32.2	67.6	32.2	85.6	85.9	7.15	7.12	195	198	0.60	0.83
-20.0	40.0	32.2	66.1	32.2	84.1	80.7	7.29	7.32	190	196	0.59	0.86

## Appendix B. Uncertainty of wind tunnel tests

The wind tunnel tests had uncertainties associated with air-flow rate measurements ( $\pm 2\%$ ) and ( $\pm 0.5^\circ\text{C}$ ) in the air temperature measurements. Using this uncertainty data [19] and uncertainty propagation tool in [18], uncertainties in the measured air-side heat transfer capacity were computed.

The measured capacity was composed of the heat lost by the refrigerant to air and heat lost by the refrigerant through the insulation to the surroundings (this part did not result in a rise in air temperature). The air was treated as air-H<sub>2</sub>O mixture with humidity ratio at the inlet computed from the specified relative humidity at the wind tunnel inlet [19]. Assuming that the humidity ratio remains constant, the air outlet properties were computed using [18]. For computation of heat lost through the insulation the values of the loss coefficient and area of the insulation were provided by the manufacturer who conducted the tests [19]. Assigning uncertainty values to the air-flow rate and air temperature measurements, uncertainty in measured capacity is computed by [19]. The result is shown below for condenser C/F#1.

Table B1 Uncertainty in measured air-side capacity (C/F#1)

T <sub>air,in</sub> (°C)	T <sub>air,out</sub> (°C)	Air-flow rate (m <sup>3</sup> /min)	Capacity calculated from ?T [W]	Reported capacity (W)	Simulated Capacity (W)
30	37.42	0.67	100.5±9.7	108	113
30	36.83	0.97	131.8±13.7	143	145
30	35.5	1.43	155.1±20.0	166	171
30	34.29	1.9	160.3±26.4	170	

±0.5°C uncertainty in air temperature measurements  
 ±2 % uncertainty in air-flow rate measurement

From Table B1, it can be seen that the computed capacity values lie within the range of experimental uncertainty. It is not clear, however, why capacity values reported in [19] do not match with capacity value computed from the air temperature measurement data from the same source. Similar observations can be made from Tables B2 and B3 for spiral and saw-tooth coils respectively.

Table B2 Uncertainty in measured air-side capacity (Spiral#1)

T <sub>air,in</sub> (°C)	T <sub>air,out</sub> (°C)	Air-flow rate (m <sup>3</sup> /min)	Capacity calculated from ?T [W]	Reported capacity (W)	Simulated Capacity (W)
30	37.3	0.67	98.9±9.7	107	114
30	36.6	0.97	128±13.7	138	141
30	35.3	1.43	149.6±20	163	168
30	34.1	1.9	153.3±26.4	168	170

±0.5°C uncertainty in air temperature measurements  
 ±2 % uncertainty in air-flow rate measurement

Table B3 Uncertainty in measured air-side capacity (S/T#1)

T <sub>air,in</sub> (°C)	T <sub>air,out</sub> (°C)	Air-flow rate (m <sup>3</sup> /min)	Capacity calculated from ΔT [W]	Reported capacity (W)	Simulated Capacity (W)
30	36.8	0.67	92.3±9.7	99.7	90.1
30	36.2	0.97	120±13.7	128	116
30	35.3	1.43	149.6±20	160.4	149
30	34.2	1.9	157±26.4	170.9	179

±0.5°C uncertainty in air temperature measurements

±2 % uncertainty in air-flow rate measurement